

APPROVAL CERTIFICATE

EASA.21J.414

Pursuant to Regulations (EU) 2018/1139 and (EU) 748/2012 and subject to the conditions specified below, the Agency hereby certifies

AMAC Aerospace Switzerland AG

**Sternengasse 18
4051 Basel
Switzerland**

as a DESIGN ORGANISATION

approved according to Part 21, Section A, Subpart J.

CONDITIONS :

1. The approval is limited to that specified in the enclosed Terms of Approval, and
2. This approval requires compliance with the procedures specified in the Design Organisation Handbook, reference AA-DOE-POE, in the latest revision, and
3. This approval is valid whilst the approved Design Organisation remains in compliance with Part 21, Section A, Subpart J.
4. Subject to compliance with the foregoing conditions, this approval shall remain valid until surrendered or revoked.

For the **European Aviation Safety Agency**,
Date of issue: 28/09/2018



Iain HIGGINS
Senior DOA Team Leader

Terms of Approval 21J.414
Issue 4, 23 September 2020

AMAC Aerospace Switzerland AG

Terms of Approval

Design Organisation Approval Certificate

EASA.21J.414

1 Scope

This Design Organisation Approval is applicable for the scope defined in Annex A for design work with regard to the airworthiness, operational suitability and environmental characteristics of the products.

2 Privileges

- a) (Reserved)
- b) (Reserved)
- c) The holder of this design organisation approval shall be entitled, within the scope of this terms of approval, and under the relevant procedures of the design assurance system:
 1. to classify changes to a type-certificate or to a supplemental type-certificate as “major” or “minor”;
 2. to approve minor changes to a type-certificate or to a supplemental type-certificate;
 3. (Reserved);
 4. (Reserved);
 5. [not applicable]
 6. to approve for certain aircraft the flight conditions under which a permit to fly can be issued in accordance with point 21.A.710(a)(2), except for permits to fly to be issued for the purpose of point 21.A.701(a)(15);
 7. to issue a permit to fly in accordance with point 21.A.711(b) for an aircraft it has designed or modified, or for which it has approved, in accordance with point 21.A.263(c)(6), the flight conditions under which the permit to fly can be issued, and where the holder of this design organisation approval itself:
 - (i) controls the configuration of the aircraft, and
 - (ii) attests conformity with the design conditions approved for the flight;
 8. [not applicable]
 9. [not applicable]

3 Obligations

The holder of this design organisation approval shall, within the scope of this terms of approval:

- a) maintain the handbook required under point 21.A.243 in conformity with the design assurance system;
- b) ensure that this handbook or the relevant procedures included by cross-reference are used as a basic working document within the organisation;
- c) determine that the design of products, or changes or repairs thereto comply with the applicable specifications and requirements and have no unsafe features;
- d) provide the Agency with statements and associated documentation confirming compliance with point (c), except for approval processes carried out in accordance with point 21.A.263(c);
- e) provide to the Agency data and information related to the actions required under point 21.A.3B;
- f) under the privilege of paragraph 2(c)(6), determine the flight conditions under which a permit to fly can be issued;
- g) under the privilege of paragraph 2(c)(7), establish compliance with points (b) and (e) of point 21.A.711 before issuing a permit to fly to an aircraft;
- h) designate data and information issued under the authority of the approved design organisation within the scope of its terms of approval as established by the Agency with the following statement: "The technical content of this document is approved under the authority of the DOA ref. EASA. 21J.414".

Date of issue: 23/09/2020



Alexandru ENACHE
Section Manager
Design Organisations & Policy Issues

Annex A

Scope of work

	TC	STC	major changes	minor changes	major repairs	minor repairs	flight conditions	permit to fly
Large aeroplane								
Avionics								
Autoflight systems		■	■	■			■	■
Communication systems		■	■	■			■	■
Diagnostic and Maintenance systems		■	■	■			■	■
Indicating, Alerting systems		■	■	■			■	■
Navigation systems		■	■	■			■	■
Recording systems		■	■	■			■	■
Surveillance systems		■	■	■			■	■
Cabin								
Cabin interiors		■	■	■			■	■
Cargo compartments		■	■	■			■	■
Electrical cabin systems		■	■	■			■	■
External schemes, placards and markings		■	■	■			■	■
Flight deck interiors		■	■	■			■	■
Electrical Systems								
Electrical generation / distribution systems		■	■	■			■	■
External lighting systems		■	■	■			■	■
Wireless transmission systems		■	■	■			■	■
Environmental Control Systems								
Air conditioning systems		■	■	■			■	■
Bleed systems		■	■	■			■	■
Ice and Rain protection systems		■	■	■			■	■
Oxygen systems		■	■	■			■	■
Pressurization systems		■	■	■			■	■
Water and waste systems		■	■	■			■	■
Flight								
Flight characteristics		■	■	■			■	■
Powerplant and Fuel Systems								
Powerplant installations				■			■	■
Structures								
Empennage		■	■	■			■	■
Fuselage		■	■	■			■	■
Support for external equipment		■	■	■			■	■

Legend:

■	Title for category of product
■	Title for design scope
■	Title for design area

■	Within scope
□	Outside scope

Limitations

Limitations common to all products and activities

1. Development and demonstration of compliance related to SW with IDAL A, B and C is excluded.
2. Development of Operational Suitability Data excludes the OSD constituent MCSD and SIMD.